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## **FLUID MECHANICS**

## CIVIL ENGINEERING

Date of Test: 22/07/2024

### ANSWER KEY >

1.	(a)	7.	(b)	13.	(d)	19.	(a)	25.	(c)
2.	(b)	8.	(d)	14.	(d)	20.	(d)	26.	(b)
3.	(b)	9.	(a)	15.	(c)	21.	(b)	27.	(a)
4.	(b)	10.	(d)	16.	(c)	22.	(a)	28.	(a)
5.	(c)	11.	(b)	17.	(a)	23.	(b)	29.	(c)
6.	(d)	12.	(b)	18.	(c)	24.	(c)	30.	(b)

### **DETAILED EXPLANATIONS**

1. (a)

Ideal fluid has zero viscosity.

2. (b)

Let *V* be the volume of fluid

$$\therefore \qquad dV = \frac{-6}{100} \times V$$

$$\Rightarrow \frac{-dV}{V} = 0.06$$

∴ Increase in pressure 
$$\Delta P = \frac{-\Delta V}{V} \times K$$
  
= 1.5 × 10<sup>9</sup> × 0.06 Pa  
= 0.090 GPa

3. (b)

In the distorted Froude model

Time ratio, 
$$T_r = \frac{L_r}{\sqrt{h_r}}$$

where  $L_r$  = horizontal length ratio,

 $h_r$  = vertical depth ratio

Time between actual high tide = 24 hours =  $T_p$ 

$$T_m = T_p \frac{L_r}{\sqrt{h_r}}$$

$$= 24 \times 60 \times \frac{\sqrt{100}}{150} = 96 \text{ min}$$

4. (b)

Buoyancy force acts through center of gravity of displaced liquid.

A large metacentric height in a vessel improves stability and makes time period of oscillation shorter.

- 5. (c)
- 6. (d)

For a laminar boundary layer

$$\frac{\delta}{x} = \frac{5}{\sqrt{R_{ex}}}$$

$$\Rightarrow \frac{\delta}{x} = \frac{5}{\sqrt{\frac{\rho v x}{\mu}}}$$



$$\therefore$$
  $\delta \alpha \sqrt{x}$ 

$$\therefore \frac{\delta_2}{\delta_1} = \sqrt{\frac{2x_1}{x_1}} \implies \delta_2 = \sqrt{2}\,\delta_1$$

### 7. (b)

The piezometric head difference depends upon the gauge reading regardless of the orientation of venturimeter whether it is horizontal, vertical or inclined.

- 8. (d)
- 9. (a)

$$h_p = \overline{x} + \frac{I_{xx}}{A\overline{x}} = \overline{x} + \frac{d^2}{12\overline{x}}$$

as 
$$(I_{xx} = \frac{bd^3}{12} \text{ and } A = bd)$$

- 10. (d)
- 11. (b)

Total head = 
$$\frac{p}{\rho g} + \frac{\alpha V^2}{2g} + z$$

Pressure head = -2 cm of mercury

$$= \frac{-2 \times 13.6}{0.75}$$
 cm of oil

$$= -0.363$$
 m of oil

Velocity, 
$$V = \frac{Q}{A} = \frac{0.07}{\frac{\pi}{4} \times 0.15^2} = 3.96 \text{ m/s}$$

Velocity head = 
$$\frac{\alpha V^2}{2g} = \frac{1.1 \times 3.96^2}{2 \times 9.81} = 0.879 \text{ m}$$

Datum head, z = 12 cm = 0.12 m

Total head = 
$$-0.363 + 0.879 + 0.12$$
  
=  $0.636$  m

### 12. (b)

For laminar flow through circular pipe,

$$u = -\frac{1}{4\mu} \frac{\partial P}{\partial x} (R^2 - r^2)$$

$$\Rightarrow \qquad 0.75 = -\frac{1}{4\mu} \frac{\partial P}{\partial x} (10^2 - 5^2) \qquad \dots(i)$$

Also, 
$$u_{\text{max}} = -\frac{1}{4\mu} \frac{\partial P}{\partial x} \left( 10^2 - 0^2 \right) \qquad ...(ii)$$

$$\frac{0.75}{u_{\text{max}}} = \frac{10^2 - 5^2}{10^2}$$

$$\Rightarrow \qquad u_{\text{max}} = 1.0 \text{ m/s}$$

13. (d)

14. (d)

Since air column weight is negligible, equating pressures on both sides of dotted line, one gets

$$p_A = p_B + (0.020 \times 0.8 \times 9.81) \sin 30^{\circ}$$

$$\Rightarrow p_A - p_B = 0.07848 \text{ kN/m}^2$$

$$= 78.48 \text{ N/m}^2 \simeq 78.5 \text{ N/m}^2$$

15. (c)

$$H \propto \frac{Q^2}{D^5}$$
 (Since H is constant)

$$Q^{2} \propto D^{5}$$

$$\left(\frac{Q_{1}}{Q_{2}}\right)^{2} = \left(\frac{D_{1}}{D_{2}}\right)^{5} \qquad (\because Q_{2} = 2Q_{1})$$

$$\left(\frac{Q_{1}}{2Q_{1}}\right) = \left(\frac{D_{1}}{D_{2}}\right)^{2.5}$$

$$0.7578 = \frac{D_{1}}{D_{2}}$$

$$\frac{D_{2}}{D_{1}} = 1.3195$$

$$\frac{A_{2}}{A_{1}} = \frac{\frac{\pi}{4} \times D_{2}^{2}}{\frac{\pi}{4} D_{1}^{2}} = (1.3195)^{2} = 1.7411$$

$$A_{2} = 1.7411A_{1}$$

:. Increase in cross-sectional area

$$= \frac{A_2 - A_1}{A_1} \times 100$$

$$= \frac{1.7411A_1 - A_1}{A_1} \times 100 = 74.11\%$$

16. (c)

Velocity, 
$$V = 3 \text{ m/s}$$

Kinematic viscosity, v = 0.9 centistokes =  $0.9 \times 10^{-6}$  m<sup>2</sup>/s

Reynolds number, 
$$Re = \frac{Vd}{v} = \frac{3 \times 0.5}{0.9 \times 10^{-6}} = 1.67 \times 10^{6}$$

Since Re > 4000,

:. Flow is turbulent

$$\therefore \frac{1}{\sqrt{f}} = 2\log_{10}\frac{r_0}{k_s} + 1.74$$

$$\Rightarrow \frac{1}{\sqrt{f}} = 2\log_{10}\frac{0.25}{0.25 \times 10^{-3}} + 1.74$$

$$\Rightarrow \frac{1}{\sqrt{f}} = 7.74$$

$$\Rightarrow f = 0.0167$$

$$\therefore h_{L} = \frac{fLV^{2}}{2gd} = \frac{0.0167 \times 300 \times (3)^{2}}{2 \times 9.81 \times 0.5} = 4.596 \text{ m} \approx 4.6 \text{ m}$$

**17. (a)** For 90° V - notch

$$Q = \frac{8}{15} C_d \sqrt{2g} H^{5/2}$$

$$\frac{Q_2}{Q_1} = \left(\frac{H_2}{H_1}\right)^{5/2} = \left(\frac{0.3}{0.15}\right)^{5/2}$$

$$= 5.65$$

18. (c)

The velocity, v = 65 m/s at y = 1 and v = 0 at y = 0 V = 65y (Linear variation)  $V = \frac{\partial \Psi}{\partial y}$   $\frac{\partial \Psi}{\partial y} = 65y$   $\Psi = 32.5 y^2$ 

19. (a)  $V = C_v \sqrt{2g(\rho_{stag} - \rho_{static})}$  $= 0.98\sqrt{2 \times 9.81 \times (3 - 0.5)} = 6.86 \text{ m/s}$ 

20. (d) At point (1, -2, 1)  $u = 2x^{2} + 3y = 2 - 6 = -4$   $v = -2xy + 3y^{3} + 3yz = 4 - 24 - 6 = -26$   $w = -\frac{3z^{2}}{2} - 2xz + 9y^{2}z = -1.5 - 2 + 36 = 32.5$   $|\vec{V}| = \sqrt{u^{2} + v^{2} + w^{2}}$   $= \sqrt{(-4)^{2} + (-26)^{2} + (32.5)^{2}} = 41.8 \text{ units}$ 

#### 21. (b)

Reynold's number, 
$$Re = \frac{\rho VD}{\mu} = \frac{1260 \times 10 \times 15}{1.5 \times 100} = 1260$$

As Re < 2000,  $\therefore$  Flow is laminar

$$\tau_0 = \frac{8\mu V}{D}$$

$$= \frac{8 \times 1.5 \times 10}{0.15} = 800 \text{ Pa}$$

#### 22. (a)

$$u = -\frac{\partial \psi}{\partial y}$$
 and  $v = \frac{\partial \psi}{\partial x}$   
 $u = -\frac{\partial (3\sqrt{2}xy)}{\partial y} = -3\sqrt{2}x$ 

$$v = \frac{\partial \left(3\sqrt{2}xy\right)}{\partial x} = 3\sqrt{2}y$$

Given,

$$\sqrt{u^2 + v^2} = 6$$

$$\sqrt{\left(-3\sqrt{2}x\right)^2 + \left(3\sqrt{2}y\right)^2} = 6$$

$$\sqrt{18x^2 + 18y^2} = 6 \qquad ...(i)$$

Given,

$$\theta = 135^{\circ}$$

And we know, slope of stream function i.e.

$$\tan \theta = \frac{v}{u}$$

$$\tan(135^\circ) = \frac{v}{u}$$

$$-1 = \frac{3\sqrt{2}y}{-3\sqrt{2}x}$$

$$x = y \qquad \dots(ii)$$

By putting equation (ii) in equation (i),

$$\sqrt{18x^2 + 18(x^2)} = 6$$

$$\sqrt{36x^2} = 6$$

$$6x = 6$$

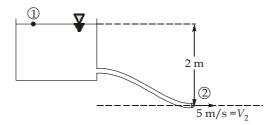
$$x = 1$$

By equation (ii),

$$y = 1$$

So, point is (1, 1).

#### 23. (b)



Applying Bernaulli's equation between (1) and (2),

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + z_2 + h_f$$
 Here, 
$$V_1 \simeq 0$$
 
$$P_1 = P_2 = 0 \quad \text{(Gauge Pressure)}$$
 
$$z_1 = 2 \text{ m (given)}$$
 
$$2 = \frac{5^2}{2g} + h_f$$
 
$$2 = \frac{25}{2 \times 10} + h_f$$
 
$$h_f = 0.75 \text{ m}$$

From Darcy-Weisbach equation,

$$h_f = \frac{f LV^2}{2gd}$$

$$\Rightarrow 0.75 = \frac{0.01 \times L \times 5^2}{2 \times 10 \times 0.05}$$

$$L = 3 \text{ m}$$

24. (c)

Ratio = 
$$\frac{\alpha_{\text{Round pipe}}}{\alpha_{\text{Parallel plates}}} = \frac{2}{1.543} = 1.3$$

25. (c)

Weight of block = 
$$25 \text{ kg}$$
  
Block dimensions =  $30 \times 30 \times 30 \text{ cm}^3$ 

Driving force along the plane,

$$F = W \sin 30^{\circ}$$
= 30 × 9.81 × 0.5  
= 147.15 N  
Shear force,  $\tau = \frac{F}{A} = \frac{147.15}{(0.3)^2} = 1635 \text{ N/m}^2$   
Contact area,  $A = 0.3 \times 0.3 \text{ m}^2$   
 $\tau = \mu \frac{dv}{dt}$ 

Also,

$$\tau = \mu \frac{dv}{dy}$$

$$\Rightarrow 1635 = 2 \times 10^{-3} \times \frac{V}{0.025 \times 10^{-3}}$$

$$\Rightarrow V = \frac{1635 \times 0.025}{2} = 817.5 \times 0.25 = 20.44 \text{ m/s}$$

### 26. (b)

Let *x* be the distance from the leading edge such that the drag force in distance *x* is one-third of the total drag force.

$$F_{Dx} = \frac{1}{3}F_{DL}$$
Now,
$$F_{Dx} = C_{D_{fx}}(Bx)\left(\frac{\rho U^2}{2}\right)$$
Also,
$$F_{DL} = C_{D_{fL}}(BL)\left(\frac{\rho U^2}{2}\right)$$

$$\therefore \frac{C_{D_{fx}}}{C_{D_{fL}}} \frac{x}{L} = \frac{F_{Dx}}{F_{DL}} = \frac{1}{3}$$
But from  $C_{D_{fx}} = \frac{1.328}{\sqrt{UL/V}}$ 

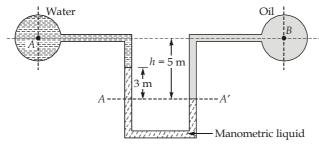
$$C_{D_{fL}} = \frac{1.328}{\sqrt{UL/V}}$$

$$\therefore \left(\frac{L}{x}\right)^{1/2} \frac{x}{L} = \frac{1}{3}$$

$$\Rightarrow \left(\frac{x}{L}\right)^{1/2} = \frac{1}{3}$$

$$\Rightarrow x = \frac{L}{9}$$

### 27. (a)



Height difference = 3 m

Specific gravity of oil = 0.9

Specific gravity of manometric liquid = 1.5

Equating pressure head at section (A-A') which is located 5 m below the centerline of pipes.

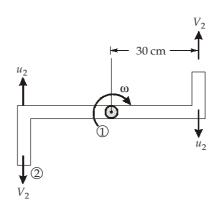
$$p_A + 3 \times 1.5 \ \rho_w g + (5 - 3) \rho_w g = p_B + 5 \times 0.9 \times \rho_w g$$

$$\Rightarrow \qquad p_A - p_B = 2.5\rho_w g - 4.5\rho_w g = -2\rho_w g$$

$$\Rightarrow$$
  $p_A - p_B = -2 \times 1000 \times 9.81 = -19620 \text{ N/m}^2$ 

$$\Rightarrow \qquad p_A - p_B = -19.62 \text{ kN/m}^2$$

### 28. (a)



Let  $\omega$  be angular velocity

∴ 
$$u_2 = \omega r$$

Given  $a = 0.8 \text{ cm}^2$ 
 $Q = 3.2 \text{ l/s} = 3200 \text{ cm}^3/\text{s}$ 

∴  $V_2 = \frac{3200}{2 \times 0.8} = 2000 \text{ cm/s}$ 
 $= 20 \text{ m/s} = \text{relative velocity of jet}$ 

Torque,  $T = -\rho Q r (u_2 - V_2) = 0$ 

⇒  $u_2 = V_2$ 

∴  $w = \frac{u_2}{r} = \frac{V_2}{r} = \frac{20}{0.3} = 66.67 \text{ radian/s}$ 

### 29. (c)

For valid potential function, Laplace equation should be satisfied i.e.  $\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$ 

$$\frac{\partial^2 \Phi}{\partial x^2} = -12y^2, \quad \frac{\partial^2 \Phi}{\partial y^2} = 12(y^2 - x^2)$$

Since  $\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} \neq 0$ , so given function is not valid since Laplace equation is based on continuity.

For irrational flow  $\frac{\partial^2 \phi}{\partial x \partial y} - \frac{\partial^2 \phi}{\partial y \partial x} = -24xy - (-24xy) = 0$ .

So, irrationality condition is satisfied but not the continuity equation.

### 30. (b)