



Rajasthan Public Service Commission
ASSISTANT ENGINEER EXAMINATION

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STRENGTH OF MATERIAL

CIVIL ENGINEERING

Date of Test : 06/11/2024

ANSWER KEY >

- | | | | | |
|---------|---------|---------|---------|---------|
| 1. (d) | 11. (c) | 21. (d) | 31. (c) | 41. (a) |
| 2. (d) | 12. (d) | 22. (d) | 32. (b) | 42. (c) |
| 3. (c) | 13. (c) | 23. (c) | 33. (d) | 43. (a) |
| 4. (b) | 14. (d) | 24. (c) | 34. (b) | 44. (a) |
| 5. (c) | 15. (d) | 25. (d) | 35. (d) | 45. (a) |
| 6. (d) | 16. (b) | 26. (d) | 36. (b) | 46. (c) |
| 7. (c) | 17. (c) | 27. (d) | 37. (d) | 47. (b) |
| 8. (c) | 18. (a) | 28. (a) | 38. (a) | 48. (a) |
| 9. (d) | 19. (a) | 29. (a) | 39. (c) | 49. (a) |
| 10. (a) | 20. (c) | 30. (c) | 40. (b) | 50. (d) |

Detailed Explanations

2. (d)

Relation between bending moment (M), shear force (V) and loading intensity (w_x).

$$w_x = \frac{dV}{dx}$$

⇒ Intensity of distributed load

= Slope of shear force diagram

$$V = \frac{dM}{dx}$$

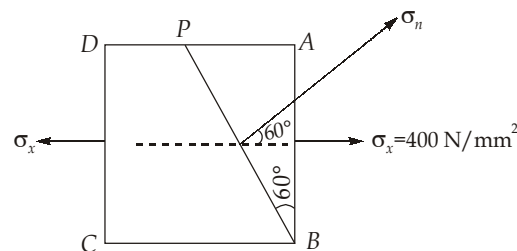
⇒ Slope of bending moment diagram at any section

= Shear force at that section.

3. (c)

Since the bar is free to expand, no stresses will be developed in the bar.

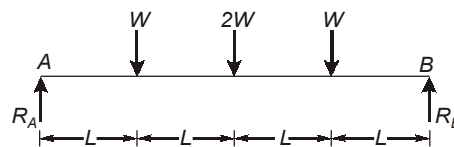
4. (b)



Normal stress on plane PB

$$\begin{aligned} \sigma_n &= \left(\frac{\sigma_x}{2} \right) + \frac{\sigma_x}{2} \cdot \cos(2 \times 60^\circ) \\ &= \frac{400}{2} + \frac{400}{2} \cdot \cos(120^\circ) = 200 + 200 \times \left(-\frac{1}{2} \right) \\ &= 100 \text{ N/mm}^2 \end{aligned}$$

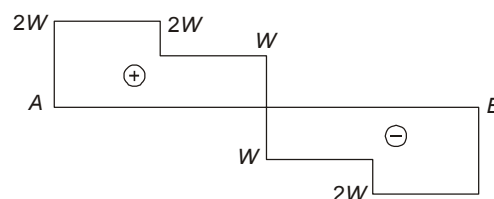
8. (c)



The reactions at the supports A and B respectively are

$$R_A = R_B = \frac{W + 2W + W}{2} = 2W$$

The SF diagram will be



The maximum shear force = $2W$.

9. (d)

In case of circular cross-section

$$\begin{aligned}\tau_{\max} &= \frac{4}{3}(\tau_{\text{avg}}) \\ &= \frac{4}{3} \times \frac{V}{A} = \frac{4}{3} \times \frac{10 \times 10^3}{\frac{\pi}{4} \times (100)^2} \\ &= \frac{16}{3\pi} \text{ N/mm}^2 = \frac{16}{3\pi} \text{ MPa}\end{aligned}$$

11. (c)

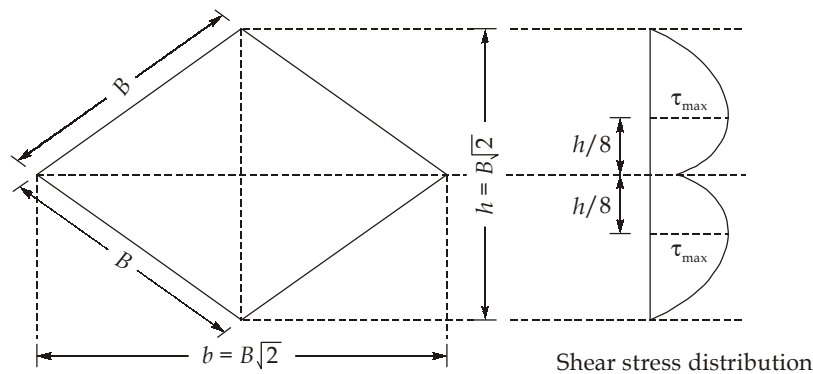
Principal stresses,

$$\sigma_1/\sigma_2 = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

Since minimum principal stress is zero, therefore

$$\begin{aligned}\left(\frac{\sigma_x + \sigma_y}{2}\right)^2 &= \left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2 \\ \tau_{xy} &= \sqrt{\sigma_x \sigma_y}\end{aligned}$$

13. (c)



The shear stress will be maximum at $\frac{h}{8} = \frac{B\sqrt{2}}{8} = \frac{B}{4\sqrt{2}}$ from neutral axis.

14. (d)

Proportionality limit shear stress

$$= \frac{300}{2} = 150 \text{ N/mm}^2$$

Maximum shear stress

$$= \frac{120 - (-30)}{2} = 75 \text{ N/mm}^2$$

$$\therefore \text{Factor of safety} = \frac{150}{75} = 2.0$$

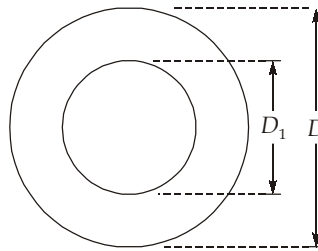
15. (d)

Equivalent moment is that moment which while acting alone produces maximum normal stress equal to the maximum principal stress due to combined action of bending and torsion.

$$\Rightarrow \frac{32M_e}{\pi D^3} = \frac{16}{\pi D^3} \cdot (M + \sqrt{M^2 + T^2})$$

$$M_e = \frac{1}{2} \cdot (M + \sqrt{M^2 + T^2})$$

16. (b)



Let P = Compressive load

$$I = \frac{\pi}{64}(D^4 - D_1^4)$$

$$\text{Stress at outer fibre} = \frac{P}{\frac{\pi}{4}(D^2 - D_1^2)} - \frac{Pe\left(\frac{D}{2}\right)}{\frac{\pi}{64}(D^4 - D_1^4)} = 0$$

$$\Rightarrow \frac{1}{D^2 - D_1^2} - \frac{e \cdot \frac{D}{2} \times 16}{(D^2 - D_1^2)(D^2 + D_1^2)} = 0$$

$$\Rightarrow e = \frac{D^2 + D_1^2}{8D}$$

17. (c)

Strain energy in torsion

$$U = \frac{1}{2} T \cdot \theta \text{ from torsion formula}$$

$$\theta = \frac{TL}{GJ}; T = \frac{\tau \cdot J}{r}; \theta = \frac{\tau \cdot L}{Gr}$$

$$U = \frac{1}{2} \times \frac{\tau \left(\frac{\pi d^4}{32} \right)}{\left(\frac{d}{2} \right)} \times \frac{\tau \cdot L}{G \cdot \frac{d}{2}}$$

$$\tau = q$$

$$\therefore U = \frac{q^2}{4G} \left(\frac{\pi}{4} \cdot d^2 \times L \right)$$

$$\therefore \frac{U}{\text{Volume}} = \frac{q^2}{4G}$$

19. (a)

$$\begin{aligned} \text{Shearing strain} &= (\epsilon_x - \epsilon_y) \sin 2\theta \\ &= (0.003 - 0.002) \sin 60^\circ \\ &= \frac{\sqrt{3}}{2} \times 10^{-3} \end{aligned}$$

21. (d)

$$R_A \times l = M$$

$$R_A = \frac{M}{l}$$

Span AC :

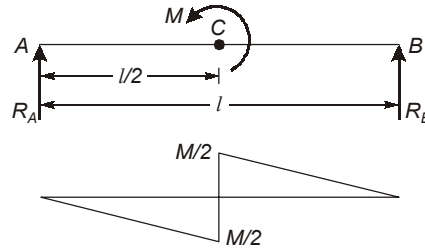
$$M_x = \frac{M}{l} x$$

$$M_C = \frac{M}{2} - M = -\frac{M}{2}$$

Span CB :

$$M_x = \frac{Mx}{l} - M$$

$$M_B = 0$$



24. (c)

$$R_A + R_B = 12 t$$

$$R_B = \frac{5 \times 2 + 2 \times 2 \times 5 + 3 \times 6}{8} = 6t$$

$$R_A = 6t$$

$$\frac{R_A}{R_B} = 1$$

26. (d)

Shafts are joined in series. So, applied torque will be same on both the shafts ($T_1 = T_2$).

From torsion formula

$$\theta = \frac{TL}{GJ}$$

Also given,

$$L_1 = L_2, G_1 = G_2$$

\therefore

$$\theta \propto \frac{1}{J}$$

$$\frac{\theta_1}{\theta_2} = \frac{J_2}{J_1} = \left(\frac{d_2}{d_1}\right)^4$$

$$(\because d_1 = 2d_2)$$

$$= \left(\frac{d_2}{2d_2}\right)^4 = \left(\frac{1}{2}\right)^4 = \frac{1}{16}$$

27. (d)

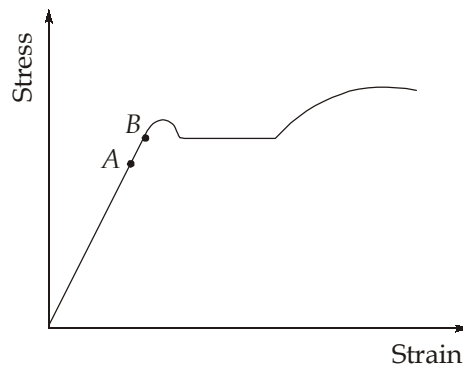
Let the reaction at the prop = R

$$\therefore \frac{5wL^4}{384EI} = \frac{RL^3}{48EI}$$

$$\Rightarrow R = 0.625 wL$$

28. (a)

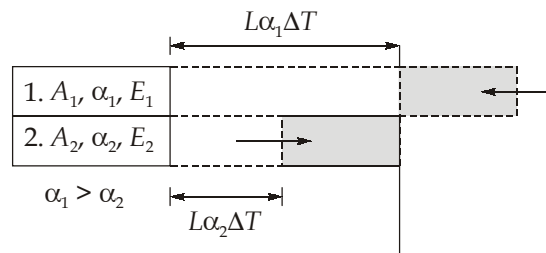
A material is said to be perfectly elastic if it regains, its original shape on removal of load.



A → Proportionality limit

B → Elastic limit

29. (a)



Hence, in case of composite bar, with temperature increase, bar having larger 'α' will have compression and that having smaller 'α' will have tension.

31. (c)

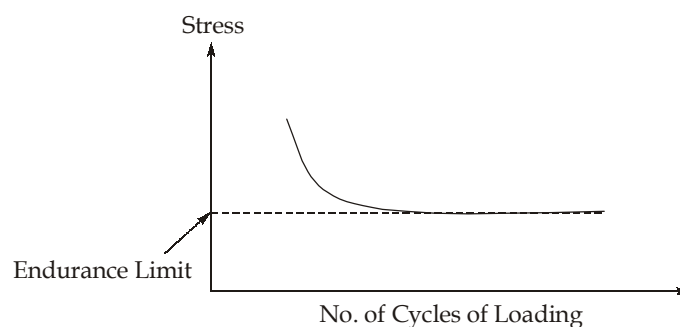
$$\frac{\tau}{R} = \frac{T}{I_p}$$

$$\Rightarrow \frac{\tau}{\left(\frac{D}{2}\right)} = \frac{T}{\left(\frac{\pi D^4}{32}\right)}$$

$$\Rightarrow T = \tau \cdot \left(\frac{\pi D^3}{16}\right)$$

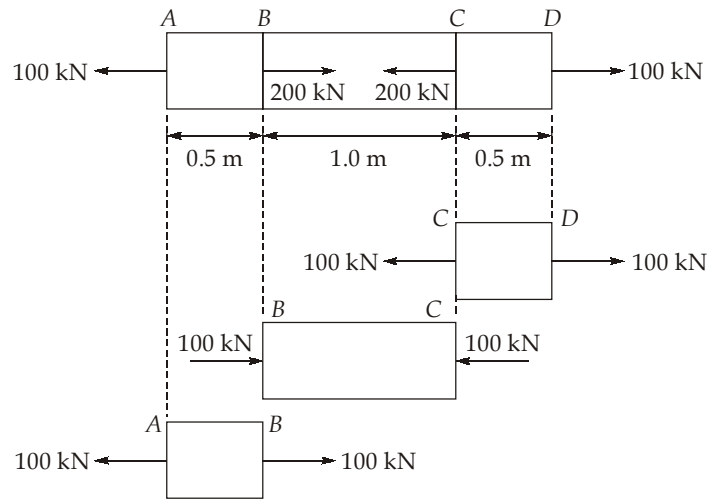
32. (b)

Endurance limit is the stress level below which even large number of stress cycles cannot produce fatigue failure.



33. (d)

FBD of bar :

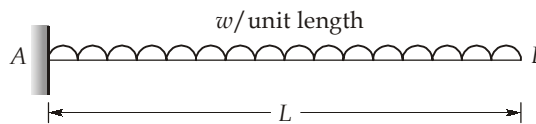


$$G = 200 \text{ GPa} = 200 \times 10^3 \text{ N/mm}^2$$

Total elongation :

$$\begin{aligned} \Delta &= \Delta_{AB} + \Delta_{BC} + \Delta_{CD} \\ &= \frac{100 \times 10^3 \times 0.5 \times 1000}{200 \times 10^3 \times 100} - \frac{100 \times 10^3 \times 1000}{200 \times 10^3 \times 100} + \frac{100 \times 10^3 \times 0.5 \times 1000}{200 \times 10^3 \times 100} \\ &= 0 \end{aligned}$$

34. (b)



$$\delta_B = \frac{wL^4}{8EI}; \theta_B = \frac{wL^3}{6EI}$$

$$\begin{aligned} \therefore \frac{\delta_B}{\theta_B} &= \frac{wL^4}{8EI} \times \frac{6EI}{wL^3} \\ &= \frac{3L}{4} \end{aligned}$$

$$\begin{aligned} \Rightarrow \frac{15 \times 10^{-3}}{0.02} &= \frac{3}{4} \times L \\ L &= \frac{60 \times 10^{-3}}{0.06} = 1 \text{ m} \end{aligned}$$

35. (d)

$$\tau = \frac{Tr}{J}$$

For solid shaft:

$$J = \frac{\pi}{32} D^4$$

$$r = \frac{D}{2}$$

For hollow shaft:

$$J = \frac{\pi}{32}(D^4 - d^4) = \frac{\pi}{32}\left(D^4 - \frac{D^4}{16}\right) = \frac{\pi}{32} \times \frac{15}{16} D^4$$

$$r = \frac{D}{2}$$

$$\therefore \frac{\tau_H}{\tau_S} = \frac{J_S}{J_H}$$

$$\therefore \frac{\tau_H}{\tau_S} = \frac{\pi/32 D^4}{\pi/32 \times \frac{15}{16} D^4} = \frac{16}{15}$$

36. (b)

The SFD does not change, while the sudden change in BMD denotes a concentrated moment at the point C. The triangular shape of BMD shows concentrated load is applied at free end.

37. (d)

Under hydrostatic loading condition, stresses at a point in all directions are equal and hence no shear stress.

Alternatively,
$$\tau = \frac{\sigma_1 - \sigma_2}{2} = \frac{50 - 50}{2} = 0$$

Thus, Mohr's circle reduces to a point.

Hence shear stress at all orientations is zero.

39. (c)

$$\begin{aligned} \text{Strain energy} &= \frac{1}{2} \times T \times \theta = \frac{1}{2} \times \frac{\tau J}{r} \times \frac{\tau L}{Cr} \\ &= \frac{1}{2} \times \frac{\tau^2 \times \pi/32 D^4 L}{CD^2/4} = \frac{1}{4C} \tau^2 \times \frac{\pi}{4} D^2 \times L \end{aligned}$$

$$\text{Strain energy} = \frac{\tau^2}{4C} \times \text{Volume}$$

42. (c)

$$\begin{aligned} \text{Power} &= T \times \omega = T \times \frac{2\pi N}{60} = \frac{1500 \times 2\pi \times 150}{60} = \frac{1500 \times 300\pi}{60} = 7500\pi \text{ W} \\ P &= 7.5\pi \text{ kW} \end{aligned}$$

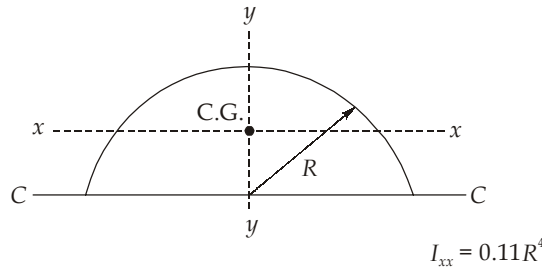
43. (a)

Hoop stress,
$$\sigma_h = \frac{Pd}{2t}$$

Longitudinal stress,
$$\sigma_l = \frac{Pd}{4t}$$

$$\therefore \frac{\sigma_h}{\sigma_l} = \frac{\frac{Pd}{2t}}{\left(\frac{Pd}{4t}\right)} = 2$$

44. (a)



Note: About diametric axis,

$$I_{CC} = \frac{\pi R^4}{8}$$

45. (a)

For hinged column, $P_{cr} = \frac{\pi^2 EI}{L^2} = 10 \text{ kN}$

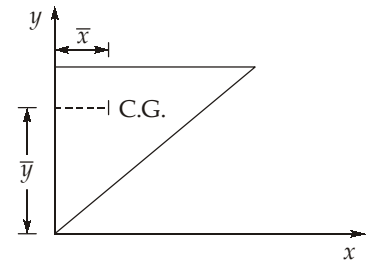
For fixed column, $P_{cr} = \frac{4\pi^2 EI}{L^2} = 4 \times 10 = 40 \text{ kN}$

46. (c)

$$\bar{x} = \frac{b}{3}$$

$$\bar{y} = \frac{2h}{3}$$

\therefore Co-ordinate of centroid $\left[\frac{b}{3}, \frac{2h}{3} \right]$.



47. (b)

We know from torsion formula

$$\theta = \frac{TL}{GJ}$$

$$GJ = \frac{TL}{\theta}$$

$GJ = \text{torsional rigidity} = T$ (for unit twist of shaft in a unit length)

48. (a)



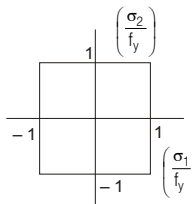
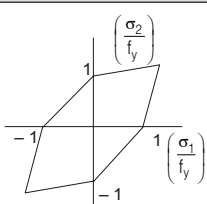
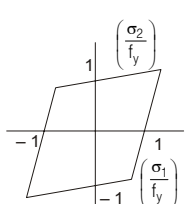
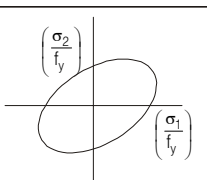
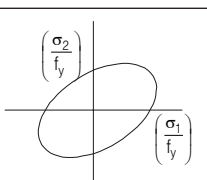
Strain energy stored in a beam due to bending

$$U = \int \frac{M^2 dx}{2EI}$$

$$M = (Px)$$

$$\therefore U = \int_0^L \frac{(Px)^2 dx}{2EI} = \frac{P^2}{2EI} \left[\frac{x^3}{3} \right]_0^L = \frac{P^2 L^3}{6EI}$$

49. (a)

(Theories of failure)	Remarks	(Theories of failure)	Remarks
(i) Maximum principal stress theory	 <p>yield locus – square.</p>	(iii) Maximum shear stress theory/Tresca, guest, coulomb theory	 <p>Yield locus – Hexagon</p>
(ii) Maximum principal strain theory/saint venant theory	 <p>Yield locus – Rhombus</p>	(iv) Maximum stress energy theory /Beltrami Haigh theory	 <p>Yield locus – ellipse</p>
		(v) Maximum distortion energy theory (Huber-Hencky-Vonmises theory)	 <p>Yield locus – ellipse</p>

50. (d)

As $\frac{dv}{dx} = w$

If SFD is parabolic (2nd degree), then the load on the beam is linearly varying distributed load (1st degree).

